

Recent Progress on Lasers for Space-Based Wind, Aerosol, and Altimetry Lidar Systems

Fibertek, Inc.
13605 Dulles Technology Drive
Herndon, VA 20171

fhovis@fibertek.com
703-471-7671

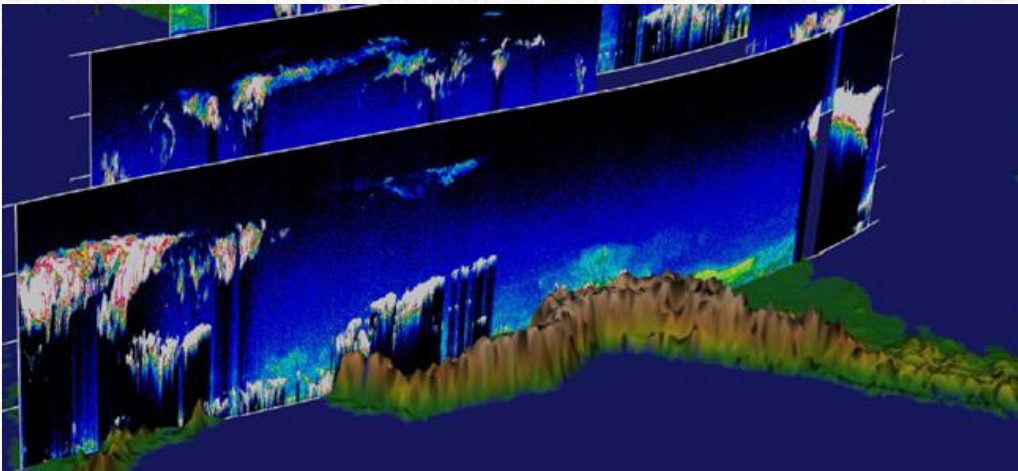
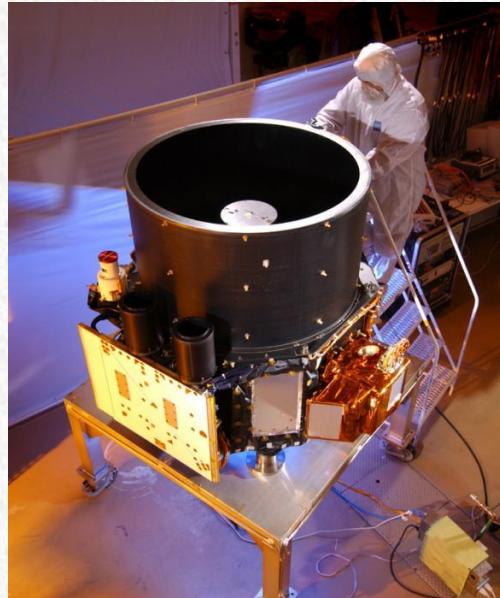




- ❖ **CALIPSO on orbit performance summary**
- ❖ **Cloud Aerosol Transport System for ISS (CATS/ISS) Status**
- ❖ **ICESat-2 Status**
- ❖ **Overview of the High Efficiency UV Demonstrator (HEUVD) Program**
- ❖ **Laser Design for ATHENA/OAWL**

CALIPSO Flight Lasers

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CALIPSO Flight Lasers

Launched 2006

Space-based aerosol
measurements

Nd:YAG
20 Hz
110 mJ @1064 and 532 nm

3 yr mission completed with
laser SN #2

Laser SN #1 operational for
additional 5 years,
projected to operate > 7 yr
 4.7×10^9 shots on orbit

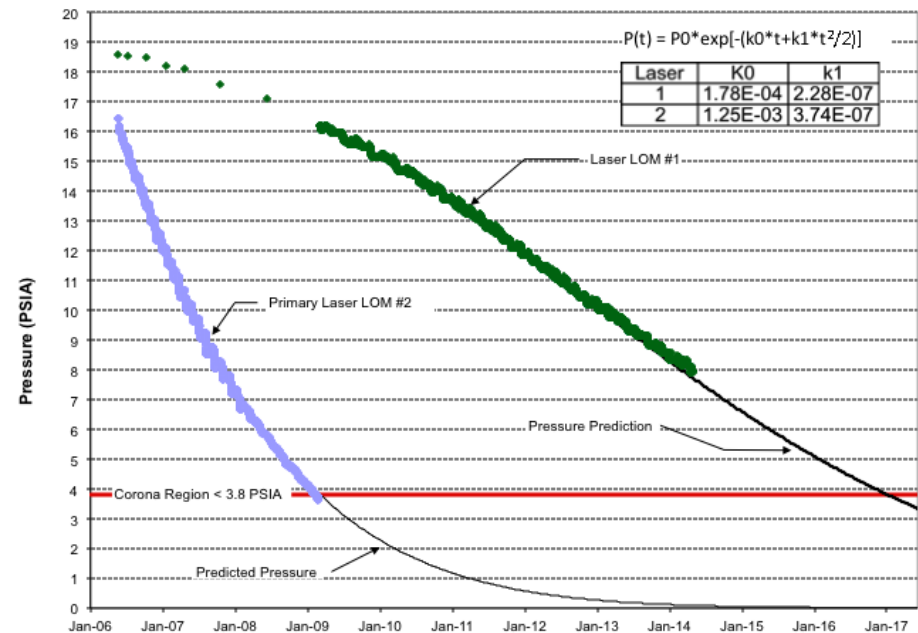
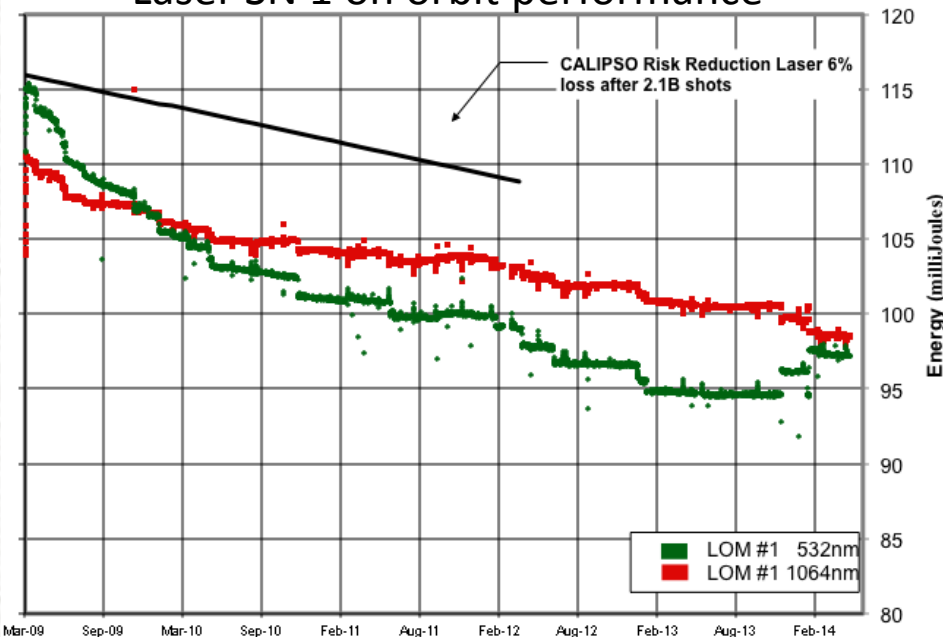
Two fully space-qualified
laser transmitters and
electronics units delivered

CALIPSO Laser On-Orbit Performance

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Laser SN 1 on orbit performance



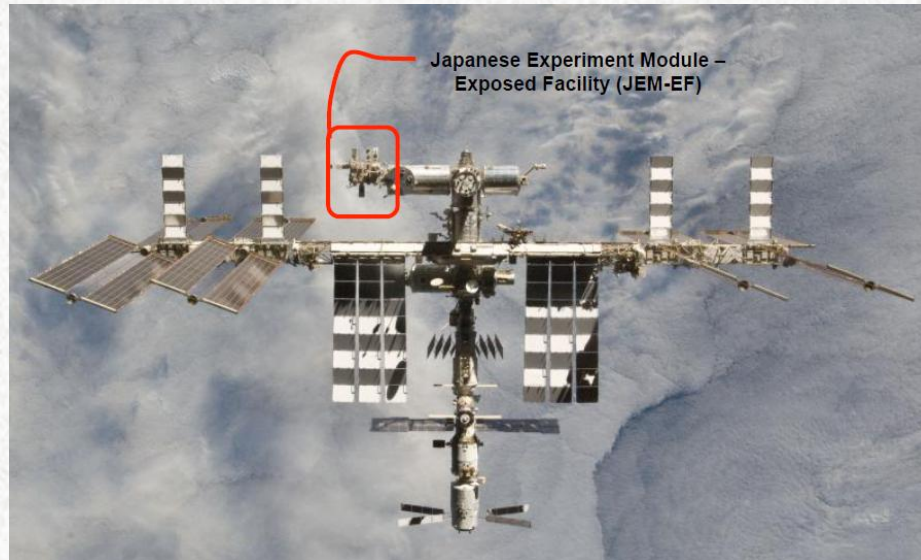
On orbit pressure profiles

- ❖ **Laser SN 2 met 3 year mission life**
 - ◆ Power decay rate was 2.2% per year
 - ◆ Lifetime limited by slow pressure leak observed before launch
- ❖ **Laser SN 1 still in operation**
 - ◆ Power decay rate has been 2.7% per year
 - ◆ Current pressure and power decay rates supports operation into 2017
- ❖ **Restart of SN 2 being evaluated**
 - ◆ Will be out of low pressure corona region by 2017

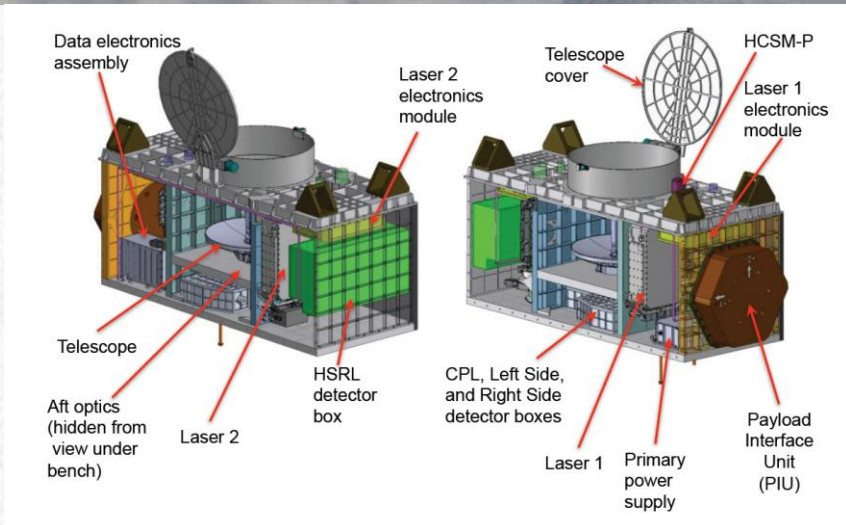
Laser SN 2

Cloud-Aerosol Transport System for ISS

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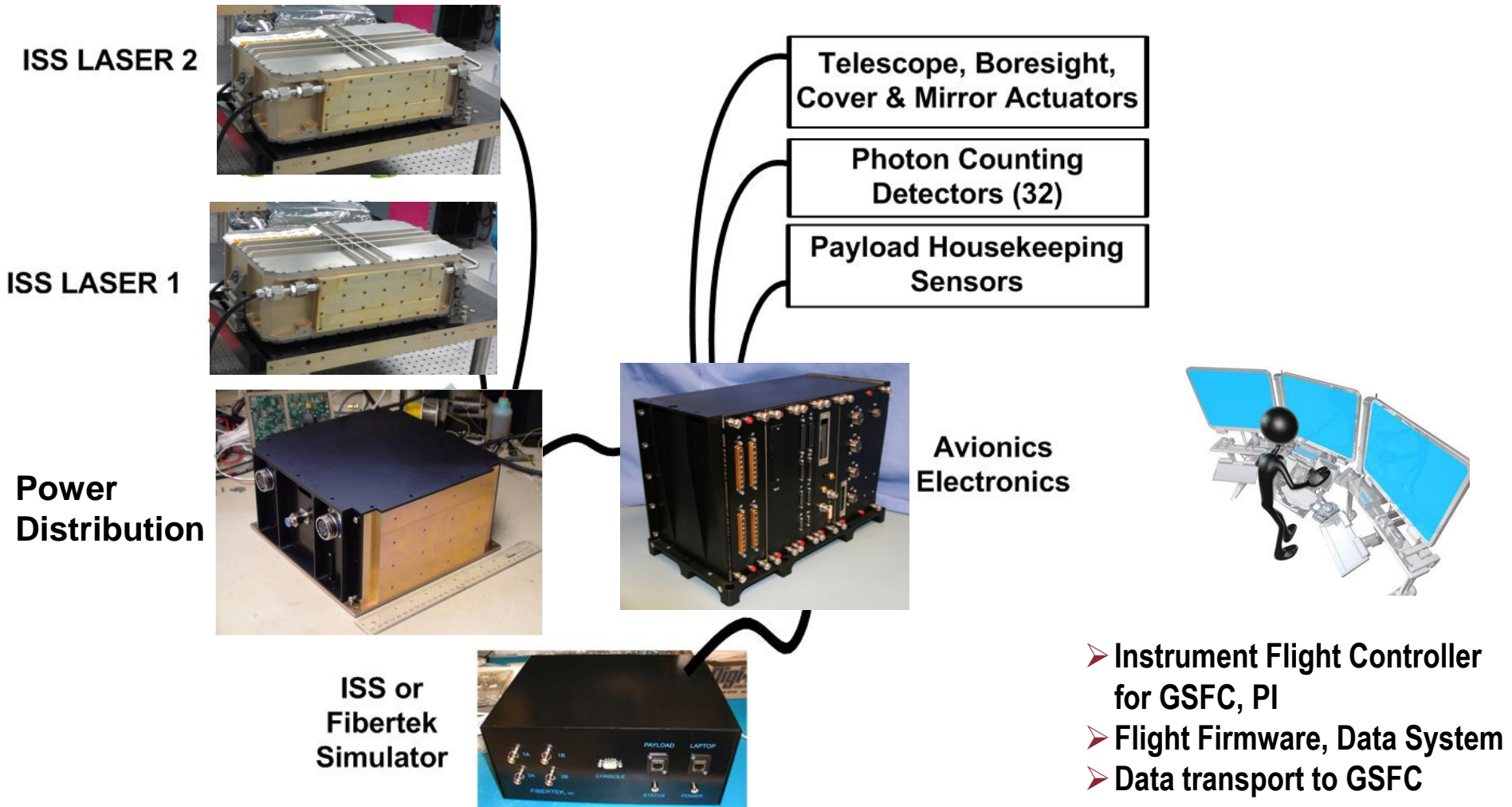


- ❖ The Cloud-Aerosol Transport System (CATS) is a LIDAR remote sensing instrument designed to provide range-resolved profile measurements of atmospheric aerosols and clouds.
- ❖ The CATS instrument uses high repetition rate lasers operating at three wavelengths (1064, 532, and 355 nm) to derive properties of cloud and aerosol layers including: layer height, layer thickness, optical depth, extinction, and depolarization-based discrimination of particle type.



Fibertek CATS-ISS Deliverables

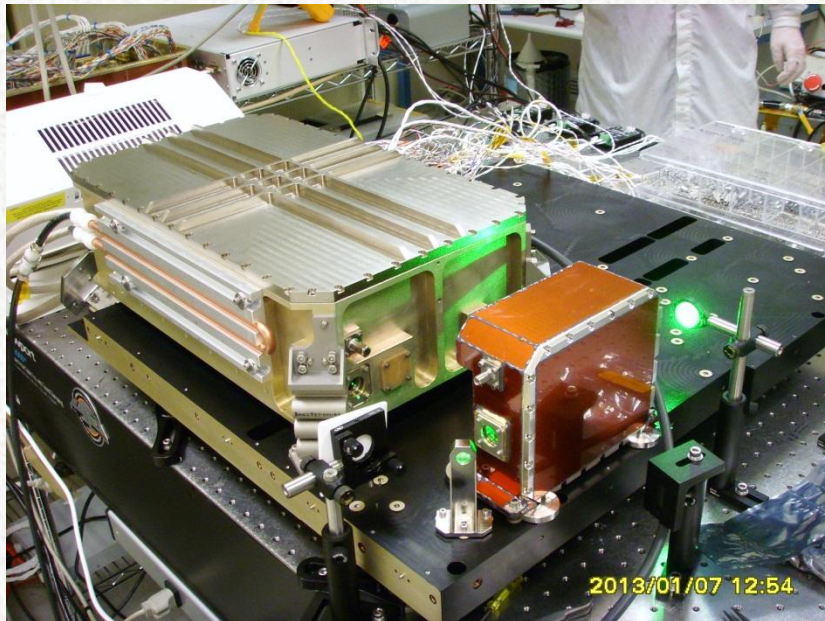
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CATS-ISS Laser Systems



Laser #2 Integrated onto the Flight Optical Bench



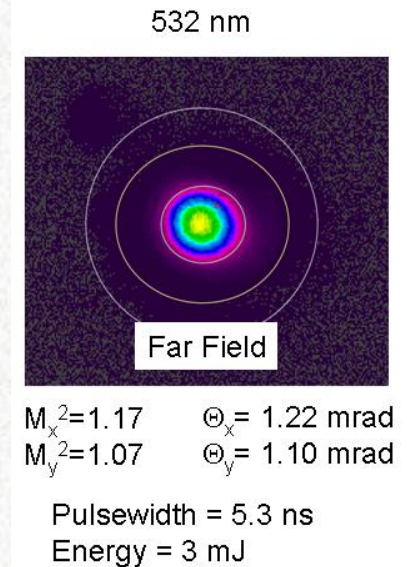
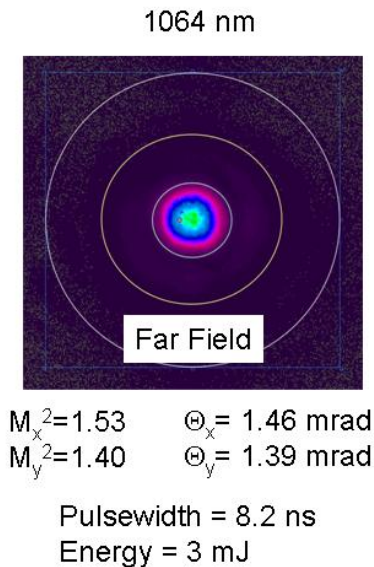
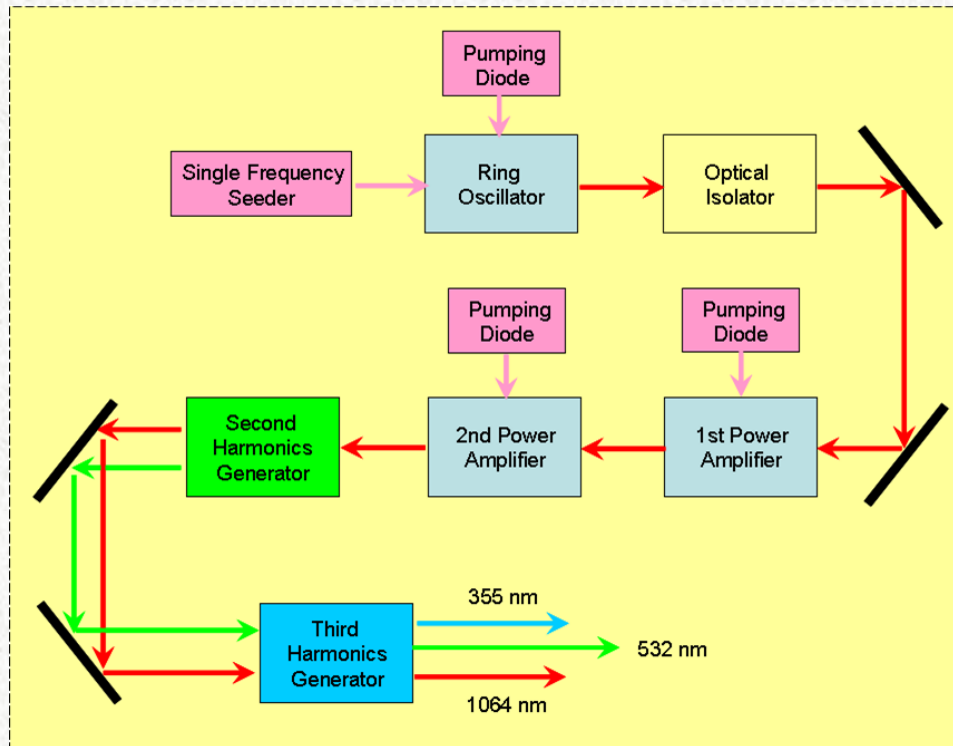
Key Performance Parameters

Parameter	Laser #1	Laser #2
Total Power (W)	27	23.3
Rep Rate (Hz)	5,000	4,000
1064 nm (W)	13	9.4
532 nm (W)	14	4.6
355 nm (W)	NA	9.3
Linewidth (pm)	0.1-0.3	Single-frequency

The combination of average power (~25 W), beam qualities ($M^2 < 1.5$), and linewidth are a significant improvement over previous flight missions

CATS-ISS Laser #2 Performance

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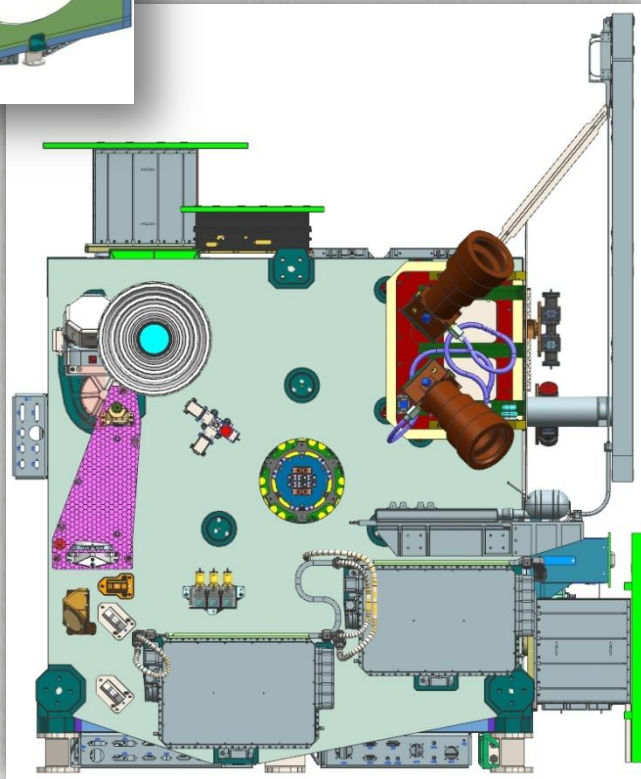
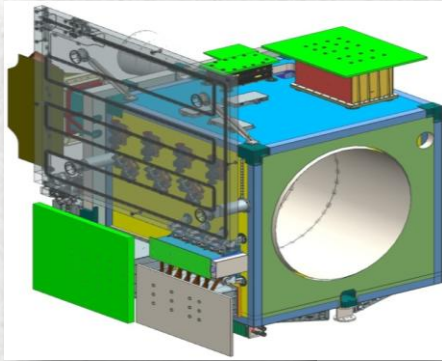
❖ CATS laser systems, 2 flight systems, were designed, built, qualified and delivered in a 2 year period.



- ❖ **Launch vehicle changed from HTV to SpaceX Dragon capsule**
 - ◆ Resulted in an increase in predicted random vibration axial loads from 4 g_{rms} to 6.5 g_{rms}
- ❖ **Full CATS-ISS payload successfully retested at 6.5 g_{rms} in May 2014**
- ❖ **CATS-ISS mission on track for a fall 2014 launch**

ICESat-2 Mission Overview

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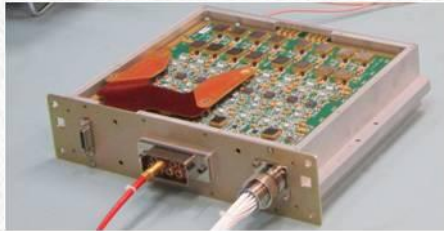
- ❖ **The ATLAS instrument is a multi-beam, micropulse, cm class laser altimeter with a dense along-track sampling of 70 cm**
- ❖ **Science Objectives:**
 - ◆ Quantifying polar ice-sheet contributions to current and recent sea-level change and the linkages to climate conditions.
 - ◆ Quantifying regional signatures of ice-sheet changes to assess mechanisms driving those changes and improve predictive ice sheet models.
 - ◆ Estimating sea-ice thickness to examine ice/ocean/atmosphere exchanges of energy, mass and moisture.
 - ◆ Measuring vegetation canopy height as a basis for estimating large-scale biomass and biomass change.
 - ◆ Enhancing the utility of other Earth observation systems through supporting measurements.

Fibertek ICESat-2 Deliverables

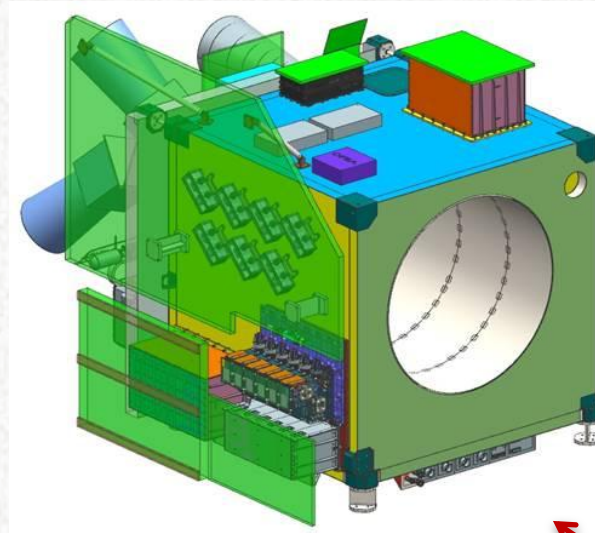
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**Photon Counting CFD
circuits (120 PMT Channels)**



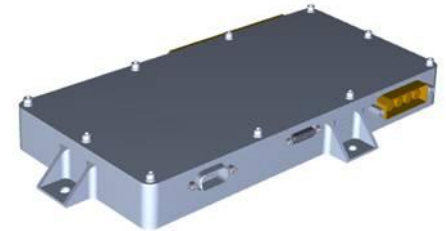
Delivery: June 2014



Lasers

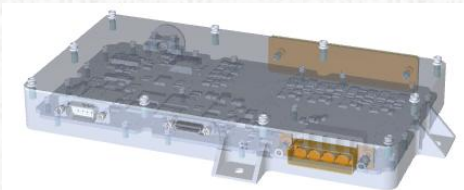


**Start Pulse Detector
Altimeter T_0 Signal
(Integrate and Test)**



Delivery: June 2014

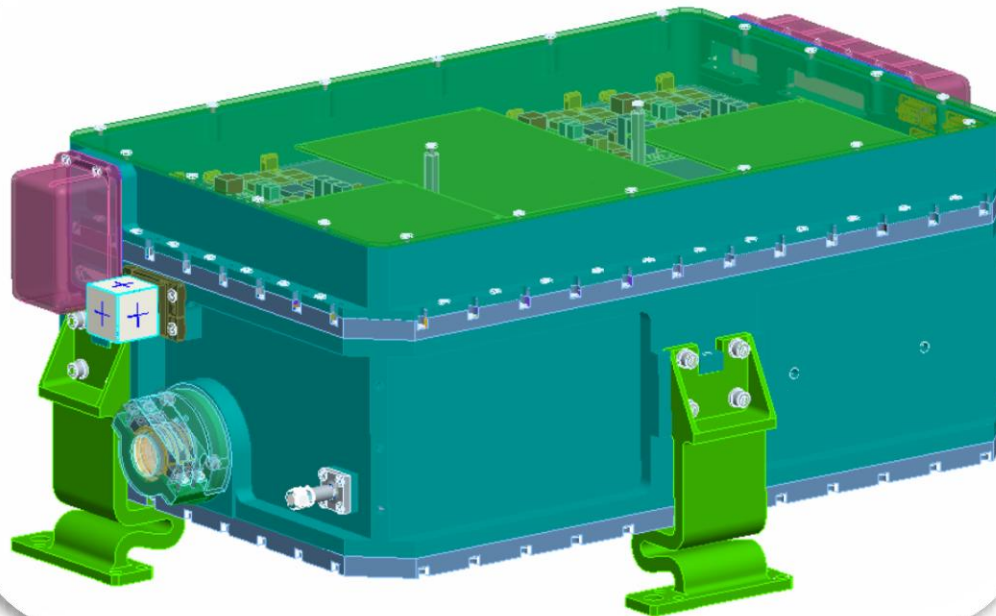
**Laser Wavelength
Control Electronics
(Integrate and Test)**



Delivery: May 2014

ICESat-2 Laser Driving Requirements

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Driving Requirements

Wall-Plug Efficiency $>5\%$

Tunable Pulse Energy $250 - 900\ \mu\text{J}$

Center Wavelength $532.\text{xxx}\ \text{nm} \pm 15\ \text{pm}$, in vacuum

Wavelength Stability & Linewidth $< 30\ \text{pm}$

Mean output wavelength tuning

Mean Pulse Width $< 1.5\ \text{ns}$

Repetition Rate $10 \pm 0.3\ \text{kHz}$

Polarization Linear 100:1

Spatial Mode $1.6 \times$ diffraction limit [$M^2 < 1.6$]

Shot-to-shot Pointing Stability $< 10.8\ \mu\text{rad}$ drift

Boresight shift during vibe testing $< 200\ \mu\text{rad}$

Lifetime 3 years + 60 days

Environmental Survival GEVS (14.1 grms qual vibe)

Flight Laser System Packaging Overview

❖ Integrated LOM/LEM with single thermal interface

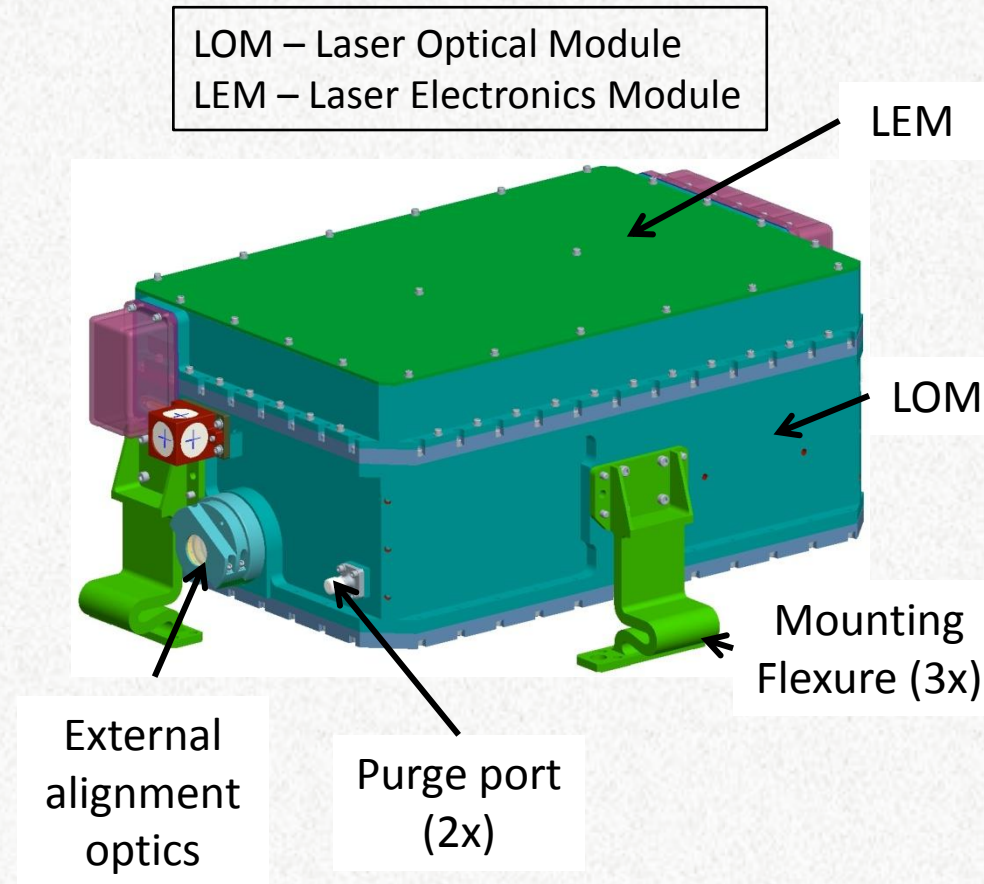
- ♦ Mass < 21.6 kg
- ♦ <50 cm x 30 cm x 15 cm
- ♦ Robust to GEVS random vibrate

❖ Wavy flexure mounts

- ♦ Filters high-frequency content of GEVS 14.1 g_{rms} random vibrate inputs
- ♦ Provides short-term & long term pointing stability

❖ Dual compartment LOM with pressure insensitive center plane

- ♦ Allows air pressurization for long optical reliability
- ♦ Short term & long term pointing stability



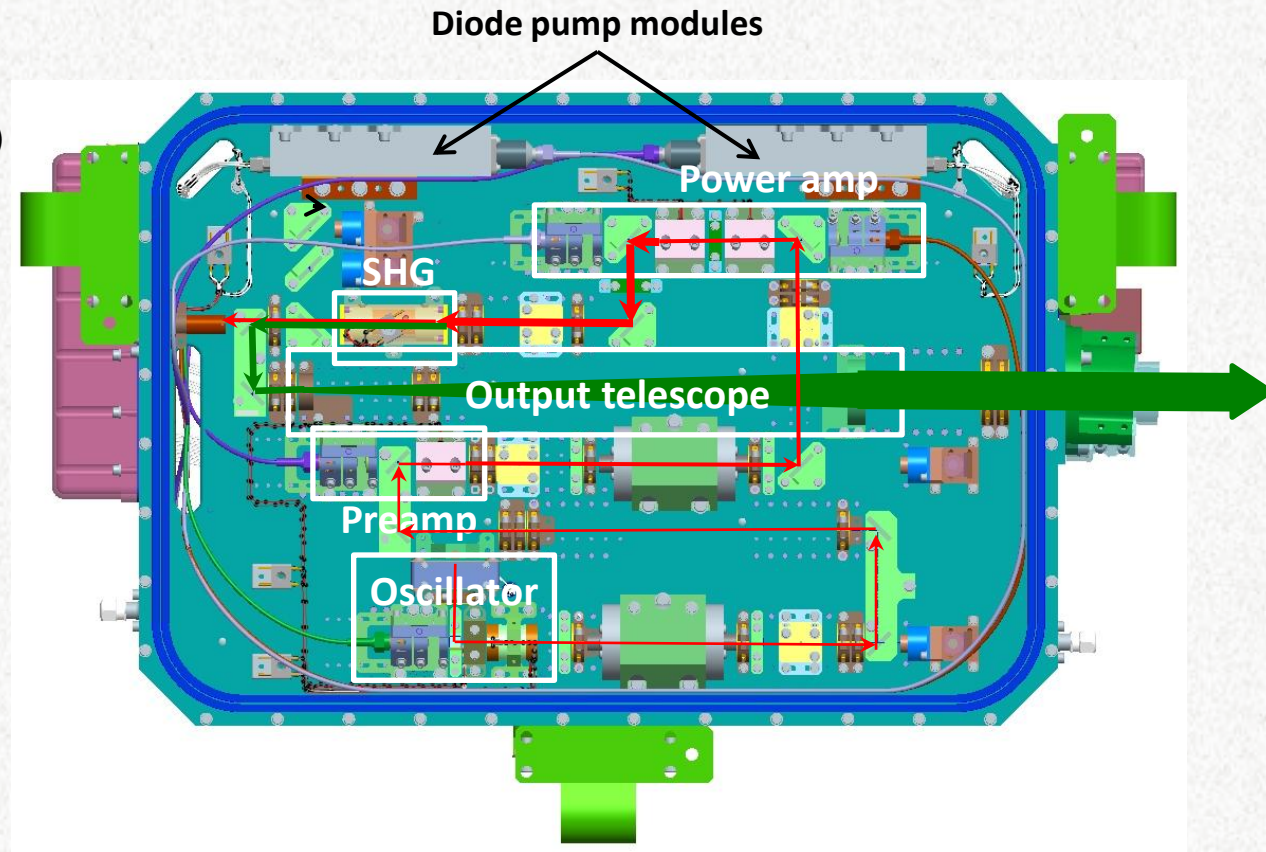
Packaging approach meets key mechanical driving requirements

Flight Optical Configuration

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- ❖ **Fiber-coupled diode pump modules**
- ❖ **Short, electro-optic (E-O) Q-switched oscillator**
 - ◆ 1.3 ns pulse width
 - ◆ 100:1 linear polarization
- ❖ **End-pumped gain medium**
 - ◆ $M^2 \sim 1.3$
 - ◆ Supports 7% wall plug efficiency @ 532 nm
- ❖ **MOPA configuration**
 - ◆ 250-900 $\mu\text{J}/\text{pulse}$
- ❖ **LBO doubled Nd:YVO₄**
 - ◆ 532 nm
- ❖ **Internal telescope**
 - ◆ 10 mm output beam



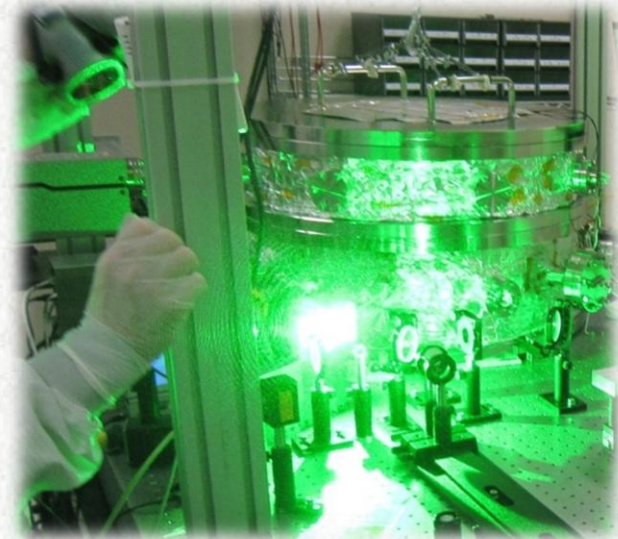
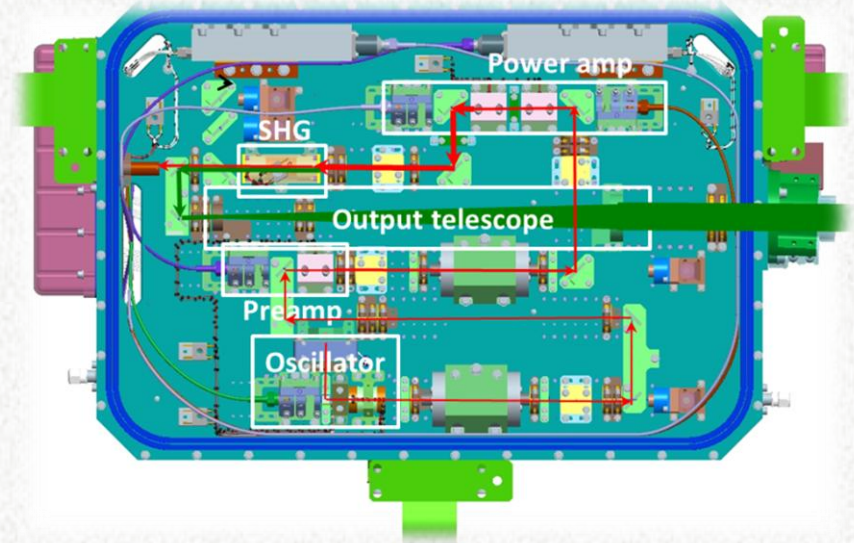
Frequency doubled Nd:YVO₄ MOPA achieves key optical driving requirements

ICESat-2 Laser Design Performance Overview

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- ❖ Fully redundant lasers and electronics
- ❖ Sealed, dual compartment laser canisters
- ❖ Air operation better understood & reduces sensitivity to trace contamination
- ❖ Master oscillator/power amplifier
 - ◆ End-pumped for high beam quality and efficiency
 - ◆ Compact EO Q-switched resonator for < 1.5 ns pulse width
 - ◆ VBG output coupler provides wavelength tuning
- ❖ End pumping achieves wall plug efficiency to 532 nm of 7%
 - ◆ 1064 nm Optical-to-Optical efficiency 35%



ICESat-2 Laser Build Status

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❖ Integration and Test Laser (qual unit)

- ◆ Laser Optics Module (LOM) fully assembled and tested
 - Meets all performance specs with margin
- ◆ Laser Electronics Module (LEM) assembled and tested
- ◆ Integrated testing of LOM and LEM is ongoing in preparation for qual testing

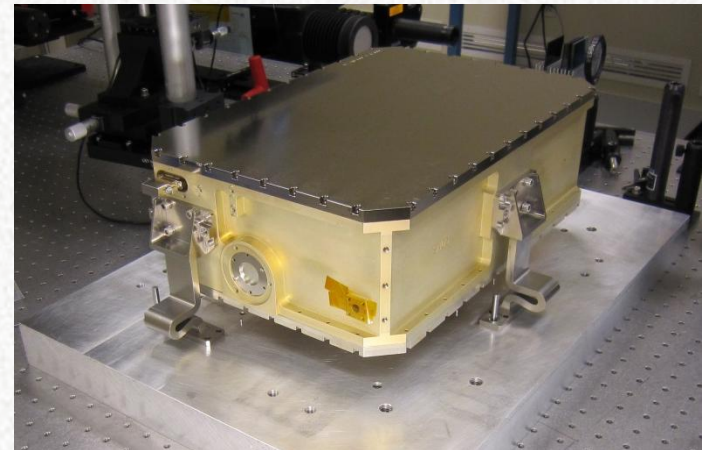
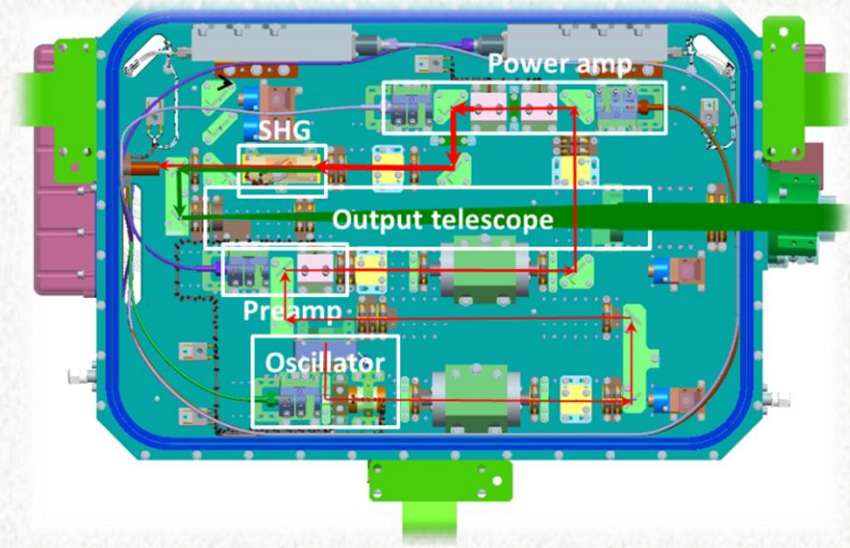
❖ Flight SN1

- ◆ LOM is built and tested through the output telescope
 - Meets all performance specs with margin
- ◆ All LEM boards are built and tested

❖ Flight SN2

- ◆ LOM is built and tested through the SHG
 - Meets all performance specs with margin
- ◆ All LEM boards are built and tested

❖ On track for Q4 2014 deliver of flight units



Flight Laser SN1 after flexure installation

High Efficiency UV Demonstrator (HEUVD) Objectives

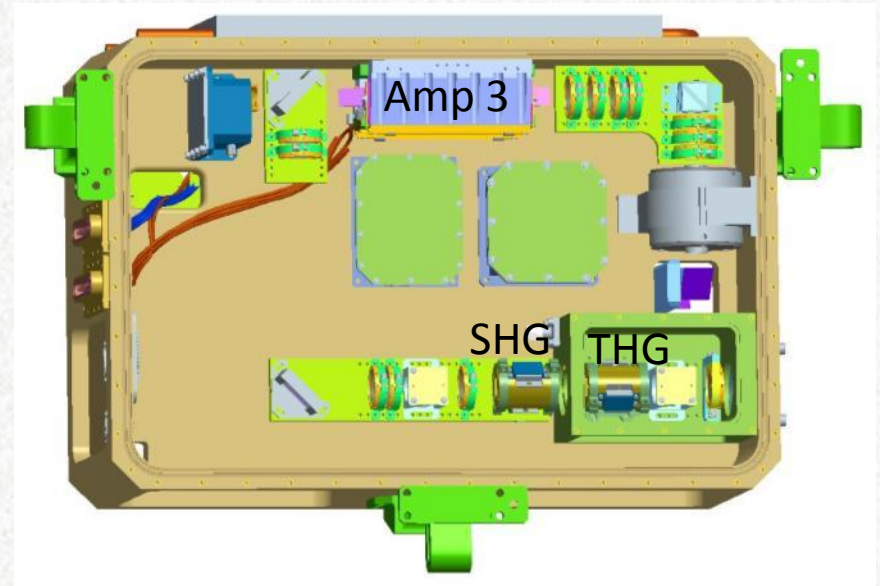
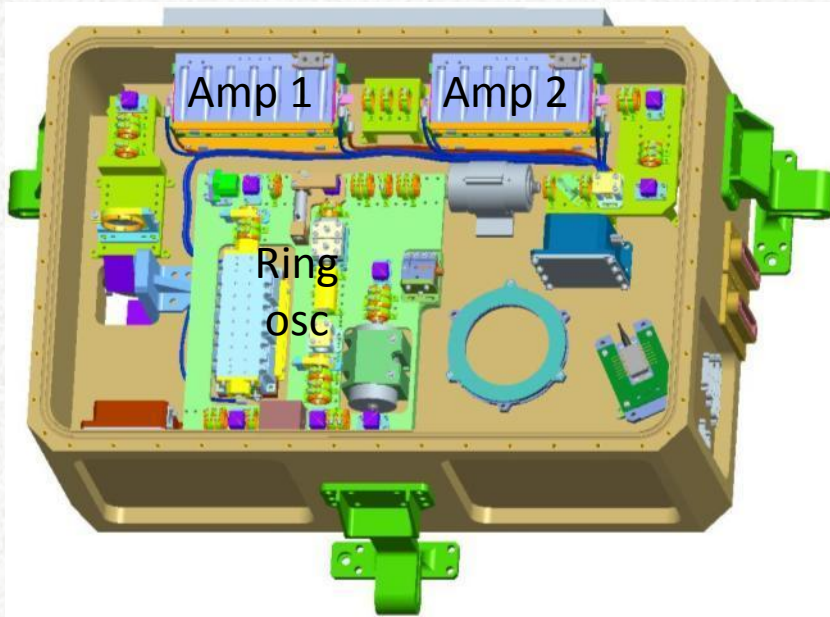
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- ❖ **ESTO funded development/risk reduction program**
- ❖ **Improved 1064 nm final power amplifier - 750 mJ/pulse @ 50 Hz with an $M^2 < 2$**
- ❖ **Purely conductively cooled Laser Optics Module (LOM) design for packaging the 50 Hz, 750 mJ 1064 nm pump laser**
- ❖ **UV conversion module design with a lifetime of $>10^9$ shots that achieves 350 mJ at 355nm**
- ❖ **Advance the design TRL from 4 to 6**
- ❖ **Execute an 8 month life test of the pump laser and UV conversion module**

HEUVD Design Overview

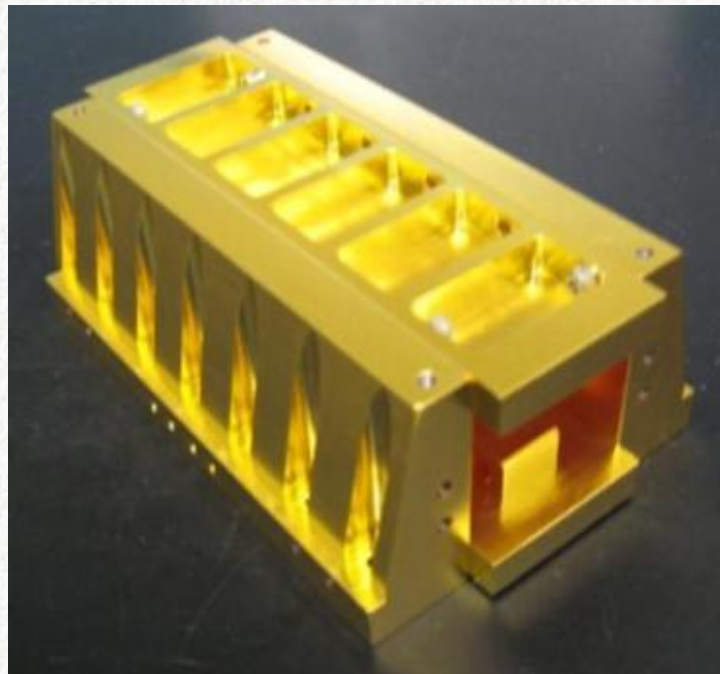
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- ❖ **Flexure-mounted, dual compartment design**
- ❖ **All amplifiers conductively cooled to a single wall**
- ❖ **Ring oscillator and first two amps lifted from HSRL-2**
- ❖ **New final power amplifier design**
 - ◆ Higher power and beam quality pump on bounce design
- ❖ **Near polymer free UV conversion module**
 - ◆ Only polymers are seal o-rings
 - ◆ Includes an internal beam expander
- ❖ **Electronics are a mix of COTS and custom designs**

HEUVD Final Power Amplifier Status

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- ❖ **Amplifier is assembled and testing is underway**
 - ◆ Probed under full pump load
 - ◆ Measured small signal gain met expectations
 - ◆ Currently quantifying focusing and impact on beam quality

Key HEUVD Milestones



- ❖ **Final boresight over pressure and thermal analyses completed in May 2014**
- ❖ **All parts ordered by late June 2014**
- ❖ **Laser assembly August-November 2014**
- ❖ **Environmental testing (vibe and TVAC) December 2014 through January 2015**
- ❖ **Lifetime testing February-September 2015**

ATHENA-OAWL Laser Design

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❖ Essentially HEUVD without the tripler

- ♦ Optimization for 150 Hz operation accomplished with modest changes to the beam conditioning optics

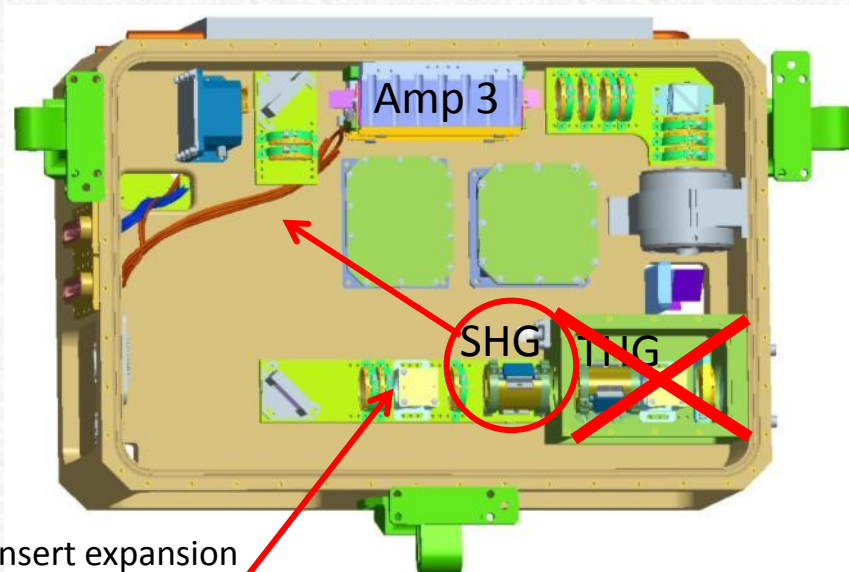
❖ Final amp will probably change to a second HSRL-2 final amp

- ♦ HEUVD final amp is overkill for required pulse energies

❖ Doubler will be moved closer to the final power amp

❖ Pressure insensitive expansion telescope will go into output leg

- ♦ Derived from ICESat-2 design
- ♦ Pressure insensitive



Insert expansion
scope here

Predicted ATHENA-OAWL laser performance	
Repetition rate (Hz)	150
Osc. 1064 nm energy (mJ)	7
Amp #1 1064 nm energy (mJ)	60
Amp #2 1064 nm energy (mJ)	150
Amp #3 1064 nm energy (mJ)	280
532 nm energy (mJ)	180
Residual 1064 nm (mJ)	100
532 nm M ²	< 2
532 nm beam diameter before expansion (mm)	3
532 nm beam diameter after expansion (mm)	22
Final beam divergence (μrad)	<150
532 nm pulse width (ns)	>25 ns